

MECE 3350U  
Control Systems

Lecture 15  
Midterm Examination Review  
and Practice Exercises

## Midterm exam - Section 21

When: Monday, Nov 12, 9:40-11:00

What: Lectures 1 to 15

Where: Room split by **first** name:

A-J	K-Z
UA2120	UL9

Prepare your formula sheet (1 page, letter size, both sides)

**Everything must be handwritten**

Your formula sheet cannot exceed 1 page (letter size), both sides.

Please write your name/student ID on the formula sheet

→ Bring a photo ID or student card.

→ Exam problems are in line with those solved in class, tutorials, and assignments.

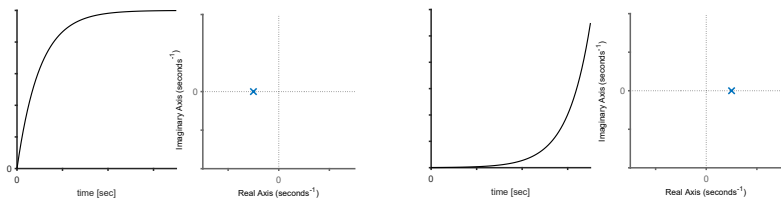
## First order transfer functions

First order functions are written in the form

$$T(s) = \frac{k}{s + \sigma}$$

where  $\tau = \frac{1}{\sigma}$  is called the time constant. The response to a unit step response is

$$y(t) = 1 - ke^{-\sigma t}$$



If  $\sigma > 0$ , the pole is on the left-half s-plane.

If  $\sigma < 0$ , the pole is on the right-half s-plane.

## Second order response

A second order system is typically represented as

$$T(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$\Rightarrow \zeta$  is the damping ratio

$\Rightarrow \omega_n$  is the undamped natural frequency

The poles of the transfer function are:

$$s_1 = \omega_n \left( -\zeta + \sqrt{\zeta^2 - 1} \right)$$

$$s_2 = \omega_n \left( -\zeta - \sqrt{\zeta^2 - 1} \right)$$

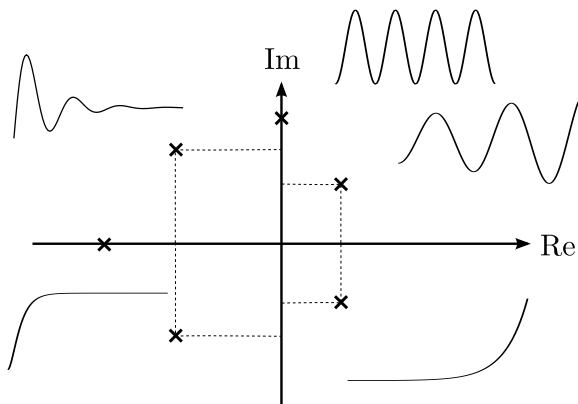
$\zeta > 1$  Overdamped system

$0 < \zeta < 1$  Underdamped system

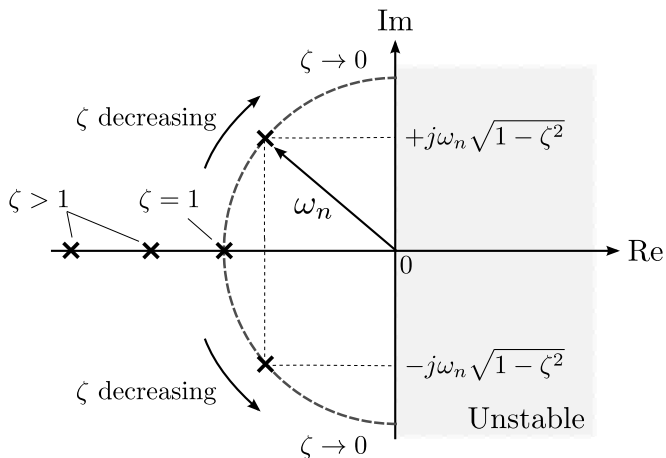
$\zeta = 1$  Critically damped system

$\zeta = 0$  Undamped system,  $\zeta < 0$  Unstable

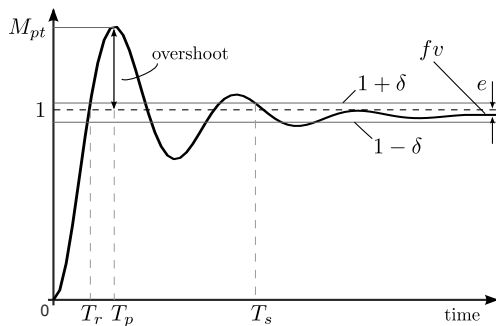
# Summary



## Summary



## Performance of feedback control systems



- Steady state error
- Rise time  $T_r$ , peak time  $T_p$ , and peak value  $M_{pt}$
- Settling time  $T_s$ :  $y(t)$  within 2% of its final value
- Percent overshoot  $P.O.$

## Performance of feedback control systems

Peak time

$$T_p = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}}$$

Magnitude at the peak time

$$M_{pt} = 1 + e^{-\zeta\pi/\sqrt{1-\zeta^2}}$$

Percentage overshoot

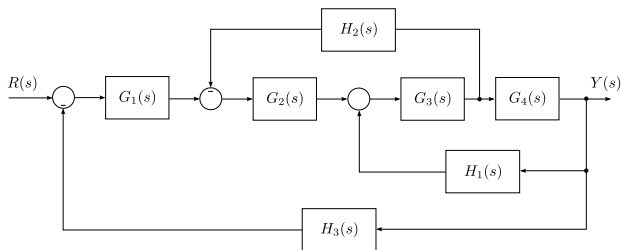
$$P.O. = 100e^{-\zeta\pi/\sqrt{1-\zeta^2}}$$

Settling time

$$T_s = \frac{4}{\zeta\omega_n} = 4\tau$$



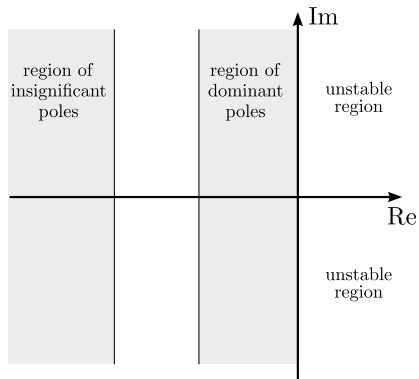
## Block diagrams



The tree fundamental operations are

- Obtain a block diagram from a transfer function
- Obtain a transfer function from a block diagram
- Simplify a block diagram

## Dominant poles



If the magnitude of the real part of a pole is at least 5 to 10 times that of a dominant pole, then the pole may be regarded as insignificant.

## The Routh-Hurwitz criterion

This criterion is a necessary and sufficient condition for stability

Order the coefficient of the characteristic equation

$$\Delta(s) = q(s) = a_n s^n + a_{n-1} s^{n-1} + \dots + a_1 s + a_0 = 0 \quad (1)$$

Into an array as follows:

$$\begin{array}{c|cccc} s^n & a_n & a_{n-2} & a_{n-4} & \dots \\ s^{n-1} & a_{n-1} & a_{n-3} & a_{n-5} & \dots \\ s^{n-2} & b_{n-1} & b_{n-3} & b_{n-5} & \dots \\ s^{n-3} & c_{n-1} & c_{n-3} & c_{n-5} & \dots \\ \vdots & \vdots & \vdots & \vdots & \\ \vdots & \vdots & \vdots & \vdots & \\ s_0 & h_{n-1} & & & \end{array} \quad (2)$$

The number of roots with positive real parts is equal to the number of changes in sign of the first column.

## The Routh-Hurwitz criterion

**Step 1:** The highest order of  $q(s)$  goes on the top-left column from  $n$  to 0.

**Step 2:** From the second column, the first two rows are the coefficients of the characteristic equation

$$\begin{array}{cccc}
 s^n & a_n & a_{n-2} & a_{n-4} & \dots \\
 s^{n-1} & a_{n-1} & a_{n-3} & a_{n-5} & \dots \\
 s^{n-2} & b_{n-1} & b_{n-3} & b_{n-5} & \dots \\
 s^{n-3} & c_{n-1} & c_{n-3} & c_{n-5} & \dots \\
 \vdots & \vdots & \vdots & \vdots & \\
 s_0 & h_{n-1} & & & 
 \end{array} \tag{3}$$

**Step 3:** Fill out the reminder rows

$$\begin{array}{cccc}
 s^n & a_n & a_{n-2} & a_{n-4} & \dots \\
 s^{n-1} & a_{n-1} & a_{n-3} & a_{n-5} & \dots \\
 s^{n-2} & b_{n-1} & b_{n-3} & b_{n-5} & \dots \\
 s^{n-3} & c_{n-1} & c_{n-3} & c_{n-5} & \dots \\
 \vdots & \vdots & \vdots & \vdots & \\
 s_0 & h_{n-1} & & & 
 \end{array}$$

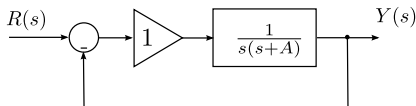
$$b_{n-1} = \frac{-1}{a_{n-1}} \left\| \begin{array}{cc} a_n & a_{n-2} \\ a_{n-1} & a_{n-3} \end{array} \right\|$$

## The root locus method

How does the location of the poles of a transfer function with characteristic equation

$$1 + kL(s)$$

change, as  $k$  goes from 0 to infinity?



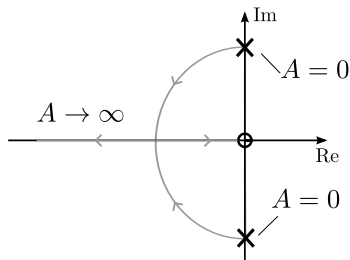
The characteristic equation is

$$1 + \frac{1}{s(s+A)} = 0$$

$$s^2 + As + 1 = 0; \rightarrow (s^2 + 1) + As = 0$$

$$\frac{(s^2 + 1)}{(s^2 + 1)} + A \frac{s}{(s^2 + 1)} = 0$$

$$1 + A \frac{s}{s^2 + 1} = 0$$



## Steps for drawing the root locus

**Step 1** Prepare the characteristic equation in the form of

$$1 + kH(s) = 0 \quad (4)$$

**Step 2** Locate the poles and zeros of  $H(s)$  in the plane

**Step 3** Locate the segments of the of the real axis that are root loci. Root loci are to the left of an odd number of poles and zeros.

**Step 4** Calculate the angle  $\theta$  and centre  $\alpha$  of asymptotes of loci that tend to infinity

$$\theta = \frac{180^\circ + 360^\circ(q - 1)}{n - m} \quad \alpha = \frac{\sum p_i - \sum z_i}{n - m}$$

**Step 5** Determine the points at which the loci cross the imaginary axis. Use Routh-Hurwitz criterion.

**Step 6** Determine the breakaway point on the real axis.

## Steps for drawing the root locus

**Step 7** Determine the angle of locus departure from complex poles and the angle of locus at arrival at complex zeros using the phase criterion.

$$q\phi = \sum \psi - \sum \phi - 180^\circ - \ell 360^\circ$$

$$q\psi = \sum \phi - \sum \psi + 180^\circ + \ell 360^\circ$$

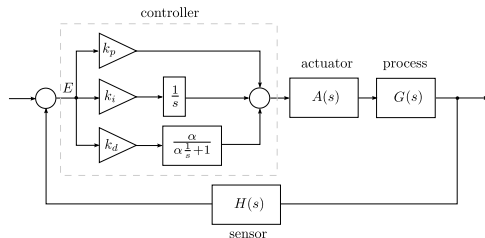
### Step 8

Complete the root locus

### Step 9

You may check you results using the Matlab function "rlocus(H);".

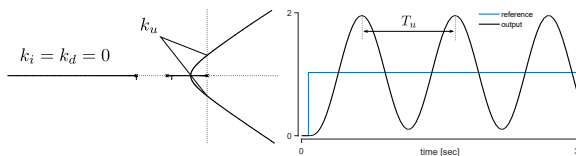
## PID controller



PID gain	Overshoot	Settling time	Steady-state error
Increasing $k_p$	Increases	Minimal impact	Decreases
Increasing $k_i$	Increases	Increases	Zero error
Increasing $k_d$	Decreases	Decreases	No impact

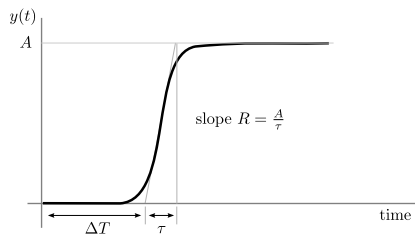


## Ziegler-Nichols PID tuning - Method 1



Controller type	$k_p$	$k_i$	$k_d$
Proportional $C(s) = k_p$	$0.5k_u$	0	0
Proportional-integral $C(s) = k_p + k_i s^{-1}$	$0.45k_u$	$\frac{0.54k_u}{T_u}$	0
PID $C(s) = k_p + k_i s^{-1} + k_d s$	$0.6k_u$	$\frac{1.2k_u}{T_u}$	$\frac{0.6k_u T_u}{8}$

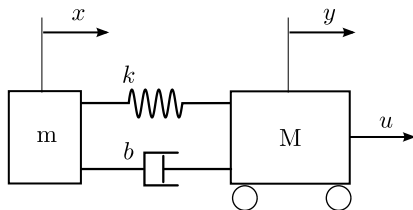
## Ziegler-Nichols PID tuning - Method 2



Controller type	$k_p$	$k_i$	$k_d$
Proportional $C(s) = k_p$	$\frac{1}{R\Delta T}$	0	0
Proportional-integral $C(s) = k_p + k_i s^{-1}$	$\frac{0.9}{R\Delta T}$	$\frac{0.27}{R\Delta T^2}$	0
PID $C(s) = k_p + k_i s^{-1} + k_d s$	$\frac{1.2}{R\Delta T}$	$\frac{0.6}{R\Delta T^2}$	$\frac{0.6}{R}$

## Exercise 73

In the system shown, a force  $u$  is applied to the mass  $M$  and another  $m$  is connected to it. The coupling between the objects is often modelled by a spring constant  $k$  with a damping coefficient  $b$ . Write the equations of motion in the Laplace domain.<sup>1</sup>

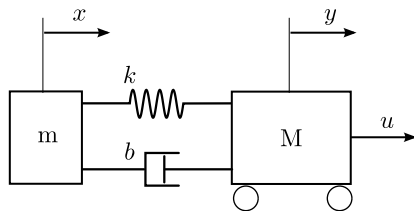


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$$\begin{aligned} m\ddot{x} &= -k(x - y) - b(\dot{x} - \dot{y}) \\ M\ddot{y} &= u + k(x - y) + b(\dot{x} - \dot{y}) \end{aligned}$$

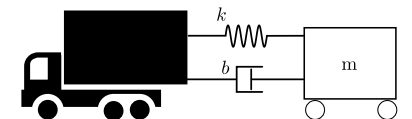
## Exercise 74

Based on the equations obtained in Exercise 68, draw a block diagram for the system of two masses.



## Exercise 75

Find the transfer function between the position of the truck and the position of the cart. <sup>2</sup>



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$$^2T(s) = (bs + k)/(ms^2 + bs + k)$$

## Exercise 76

Without computing the inverse transformation, sketch the temporal response of the following transfer functions to a step input. Specify the steady state value. Verify your plots using Matlab.<sup>3</sup>

$$T(s) = \frac{1}{s^2 + s + a}$$

$$D(s) = \frac{1}{s^2 + 5s + 1}$$

$$R(s) = \frac{1}{s^2 + 2}$$

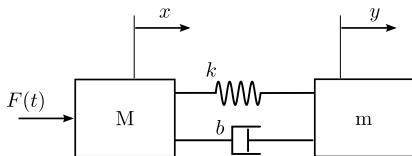
$$H(s) = \frac{50}{s^2 + 15s + 50}$$

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<sup>3</sup>Solutions can be found using Matlab

## Exercise 77

A robot includes significant flexibility in the arm members with a heavy load in the gripper. A two-mass model of the robot is shown in the figure. Find the transfer function  $Y(s)/F(s)$ .<sup>4</sup>

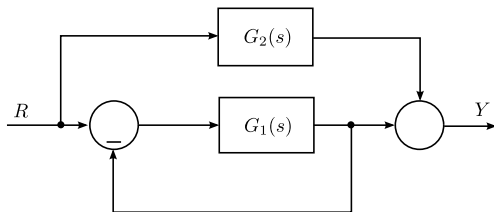


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$${}^4 T(s) = \frac{\frac{1}{mM}(bs+k)}{s^2 \left[ s^2 + \left(1 + \frac{m}{M}\right) \left( \frac{b}{m}s + \frac{k}{m} \right) \right]}$$

## Exercise 78

Find the transfer function  $Y(s)/R(s)$  for the block diagram shown.<sup>5</sup>



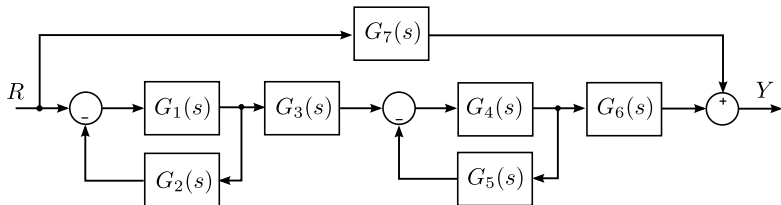
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$${}^5 T(s) = \frac{G_1}{1+G_1} + G_2$$



## Exercise 79

Find the transfer function  $Y(s)/R(s)$  for the block diagram shown.<sup>6</sup>

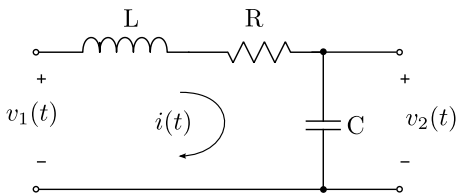


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$${}^6 T(s) = G_7 + \frac{G_1 G_3 G_4 G_6}{(1+G_1 G_2)(1+G_4 G_5)}$$

## Exercise 80

Consider the LRC circuit shown.



Find the following:

- The time domain equation relating  $i(t)$  and  $v_1(t)$
- The time domain equation relating  $i(t)$  and  $v_2(t)$
- The transfer function  $V_2(s)/V_1(s)$
- The circuit damping ration and the natural frequency
- The value of  $R$  that results in  $v_2(t)$  having an overshoot no more than 25% for an unit step of  $v_1(t)$ . Take  $L = 10$  mH,  $C = 4\mu\text{F}$ .

## Exercise 80 - continued

Solution

$$(a) v_1(t) = L \frac{di(t)}{dt} + Ri + \frac{1}{C} \int i(t) dt$$

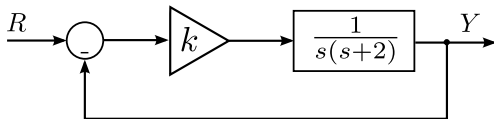
$$(b) v_2(t) = \frac{1}{C} \int i(t) dt$$

$$(c) \frac{V_2(s)}{V_1(s)} = \frac{1}{s^2 LC + sRC + 1}$$

(e) For 25% overshoot,  $\zeta = 0.4$  and thus  $R = 40\Omega$

## Exercise 81

For the unit feedback closed-loop system shown, specify the proportional controller gain  $k$  so that the output  $y(t)$  has an overshoot of no more than 10% in response to a unit step.<sup>7</sup>

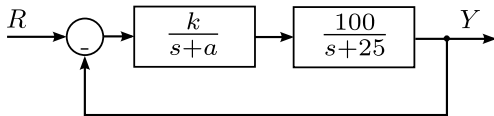


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<sup>7</sup> $\zeta \geq 0.591$ , thus  $0 < k \leq 2.86$

## Exercise 82

For the unit feedback closed-loop system shown, specify the proportional controller gain  $k$  and the location of the pole  $a$  so that the output  $y(t)$  has an overshoot of no more than 25%, and a settling time of no more than 0.1 sec in response to a unit step.<sup>8</sup>



Verify your results using Matlab.

<sup>8</sup> $\zeta \geq 0.4037$ ,  $\omega_n \approx 114$ , thus  $a = 67.1$ , and  $k \approx 113$ .

## Exercise 83

Two closed-loop transfer functions are given below.

$$\frac{Y(s)}{R(s)} = \frac{2}{s^2 + 2s + 2}$$

$$\frac{Y(s)}{R(s)} = \frac{2s + 6}{2(s^2 + 2s + 2)}$$

In each case, provide estimates of the rise-time, settling time, and percent overshoot to a unit input in  $r(t)$ .<sup>9</sup>

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<sup>9</sup> $t_r = 1.27$   $t_s = 4.6$  sec,  $M_p = 5\%$ ,  $\zeta = 0.5$

## Exercise 84

Using Routh's stability criterion, determine how many roots with positive real parts the following equations have.<sup>10</sup>

(a)  $s^4 + 8s^3 + 32s^2 + 80s + 100 = 0$

(b)  $s^4 + 2s^3 + 7s^2 - 2s + 8 = 0$

(c)  $s^3 + s^2 + 20s + 78 = 0$

(d)  $s^4 + 6s^2 + 25 = 0$

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<sup>10</sup>Use the provided Matlab code to check your answers.

## Exercise 85

The transfer function of a typical hard drive system is given by

$$G(s) = \frac{k(s + 4)}{s(s + 0.5)(s + 1)(s^2 + 0.4s + 4)}$$

Using Routh's stability criterion, determine the range of  $k$  for which this system is stable when the characteristic equation is  $1 + G(s) = 0$ .<sup>11</sup>

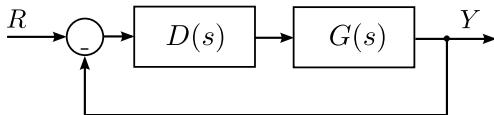
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<sup>11</sup> $0 < k < 0.78$



## Exercise 86

Consider the following closed-loop system



where

$$G(s) = \frac{1}{s}, \quad D(s) = \frac{k}{s+p}$$

Find  $k$ , and  $p$  so that the system has a 10% overshoot to a step input and a settling time of 1.5 sec.<sup>12</sup>

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<sup>12</sup> $\zeta = 0.7, k = 20.25, p = 6.3$

## Exercise 87

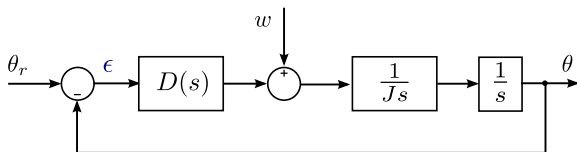
Consider the satellite altitude controller shown where the parameters are

$J = 10$  space craft inertia ( $\text{N}\cdot\text{m}\cdot\text{sec}^2/\text{rad}$ ),

$\theta_r$  reference satellite altitude (rad)

$\theta$  actual satellite altitude (rad)

$w$  disturbance torque ( $\text{N}\cdot\text{m}$ )



Continued next slide

## Exercise 87 - continued

(a) Use propositional controller ( $D(s) = k$ ) and evaluate the stability of the system.

Determine the steady-state value of  $\theta$  for the following scenarios

(b) Using PD control and a unit step reference input.

(c) Using PD control and a unit disturbance step input.

(d) Using PI control control and a unit step reference input.

(e) Using PI control and a unit disturbance step input.

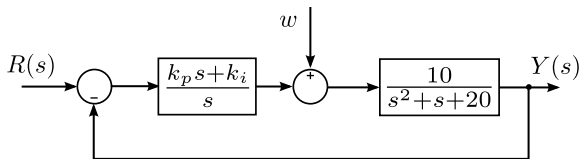
(f) Using PID control and a unit step reference input.

(g) Using PID control and a unit disturbance step input.

(a) The system is unstable, (b) 1 rad, (c)  $1/(k_p)$ , (d-e) the system is unstable, (f) 1 rad, (g) 0

## Exercise 88

Consider the system shown with PI control<sup>13</sup>

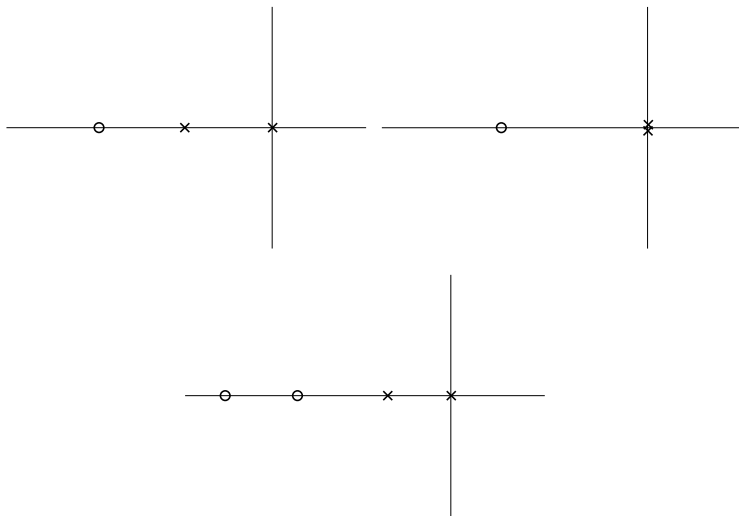


- (a) Determine the transfer function from  $Y(s)/R(s)$  and  $Y(s)/W(s)$ ,  
(b) Use Routh's criterion to find the range of  $k_p$  and  $k_i$  for which the system is stable.

<sup>13</sup>(b),  $k_i > 0$  and  $k_p > k_i - 2$

## Exercise 89

Sketch the root locus



## Exercise 89 - continued

To verify your results using Matlab, copy and past the following code

```
s = tf([1 0],[1]);
```

```
figure
```

```
rlocus((s+10)/(s*(s+5)))
```

```
figure
```

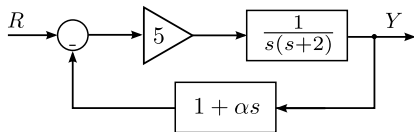
```
rlocus((s+5)/(s*s))
```

```
figure
```

```
rlocus((s+10)*(s+8)/(s*(s+4)))
```

## Exercise 90

Sketch the root locus with respect to the parameter  $\alpha$ , estimate the closed-loop pole locations, and sketch the corresponding step responses when  $\alpha = 0$ ,  $\alpha = 0.5$  and  $\alpha = 2$ . Use Matlab to check the accuracy of your approximate step responses <sup>14</sup>.



<sup>14</sup>The characteristic equation is  $1 + \alpha \frac{s}{s^2 + 2s + 5}$

## Exercise 91

A control system for positioning the head of a floppy disk drive has the closed-loop transfer function

$$T(s) = 11.1 \frac{s + 18}{(s + 20)(s^2 + 4s + 10)}.$$

Plot the poles and zeros of this system and discuss the dominance of the complex poles. What percentage overshoot for a step input do you expect? Compare the results with the actual response using Matlab.<sup>15</sup>

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<sup>15</sup>Dominant poles: 7.69%, actual overshoot 8%



## Exercise 92

A unit feedback control system has the loop transfer function

$$L(s) = k \frac{s^2 + 10s + 30}{s^2(s + 10)}.$$

We desire the dominant roots to have a damping ratio of  $\zeta = 0.707$ . Find the gain  $k$  when this condition is satisfied. <sup>16</sup>

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<sup>16</sup> $k = 16$

## Next class...

- Midterm examination